

Form 619.

1623493 A.C. 2 BIRD 3H

PRINCIPLES OF FLIGHT
AND
METEOROLOGY.

ROYAL AIR FORCE.

I FLIGHT.
10 COURSE.

Notebook for use in Schools.

91/2250. Wt. 6865. 500M. Bks. 12/41. J.D. & Co. Ltd.

Principles of Flight

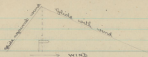
Air Speed and Ground Speed

22nd Sept 1944

The lift on an aircraft is obtained by the motion of the aircraft thro' the air. To maintain it in the air, it must have a certain speed relative to the air, called air speed (A.S). The speed relative to the ground (G.S) is not required.

Effect of wind

There is always a "head wind" when flying, even when the wind relative to the earth is from one side. The airflow or relative wind over the aircraft is still from ahead due to air speed. The earth wind and ground speed are only taken into account in landings, take off and then we reduce the ground speed by flying into wind. In a climb or glide we may need the actual path relative to the ground so we consider the earth wind - the path against the wind being steep. Motion relative to the ground seems to be important only when we can see the ground. We may appear to have sufficient speed if flying down wind and the airspeed may be very low. Illusions may be dangerous e.g. when turning with or into wind.



Change in wind speed

If an aircraft is flying into wind and the wind speed suddenly increases the aircraft tends to go on at the same ground speed as before due to its inertia - so the air speed increases. If the wind speed suddenly decreases the air speed will suddenly decrease and may result in loss of control.

Wind Gradient

This is the change in wind speed with height. Fifty feet to a hundred feet from the ground friction with the earth's surface slows up the wind so that the wind at fifty feet may be far less than the wind at 200 ft. This is important in landings and take-offs.

Aerofoils

The shape of the wings or aerofoils is so provide sufficient force to support the aircraft. They are designed so as to give a smooth airflow and to

meet the airflow at a small angle of attack. The angle of attack is the angle between the chord line (usually a line from leading to trailing edge) and the relative wind which is the reciprocal of the flight path.



Airflow



The airflow for a small angle of attack is smooth with an upwash in front and a downwash behind the wing. The wing pushes the air downwards and in doing so experiences an upward reaction from the air. The streamlines are close above giving increased speed of air and further apart below giving decreased speed.

Distribution of

Pressure



The sum of the energy due to the speed and the static pressure remains constant for a streamlined airflow. Therefore if the speed is decreased (below the airfoil) the pressure increases and if the speed is increased above the airfoil the pressure decreases. The total average difference is about 2.3 lb/sq. inch giving about 12.3 lb/sq. foot .

Center of pressure



R.W.

The effects of the pressures distributed round the airfoil can be combined into a resultant force at right angles to the chord line thus the center of pressure.

This resultant force can be resolved at right angles to the relative wind giving lift and parallel to the relative wind giving drag. Lift is used in lifting the aircraft and overcoming the weight. Drag results forward motion. In an efficient modern wing L is greater than D 11.

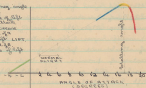
Lift and drag depend on:-

- (a) The shape of the cross section of the wing.
- (b) The angle of attack on the wing
- (c) The area of the wing

- (a) The density of the air
- (b) Air speed (as the square)

The effect of increasing the angle of attack

- Rise lift at a neg. angle of attack
- Steady increase of lift with angle of attack
- Break off on increase of lift with α 20°
- No increase of lift with increased α 20°
- Rapid decrease of lift



0°



15° (CRITICAL ANGLE)



20° (STALL)

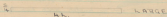
Downing more lift (and drag) from denser air.

Downing creates wing area great lift and drag.

Speed increased speed increases lift and drag. But the speed goes down then the lift and drag.

Stalling increase in angle of attack alters the coefficient and also alters the nature of flow. If center of pressure moves forward tends to increase the angle of attack and further forward movement of the center of pressure. This tends to raise nose of the machine and give instability. If after maximum lift has still decrease in center of pressure and leading edge causes nose to fall and decrease the angle of attack, no tends automatically to readjust the balance.

Aspect Ratio is the ratio of span to chord.



Increase in aspect ratio increases lift and to lesser extent drag. Structurally high aspect ratio is difficult

to make compromise.

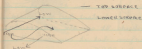
Billow and Monoplanes

Monoplanes not efficient because of interference in a relation of upper with lower wing causing a loss of lift to both wings. Can be overcome by a certain extent by stagger i.e. one wing being placed in front of the other.

Advantage lies in structural efficiency, lightness.

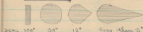
Drag reduces speed so must be reduced as much as possible.

Wing Drag: Induced drag

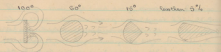


Induced is a result of a decreased pressure above and increased pressure below wing causing vortices i.e. eddies at the wing tips. is reduced by taper. is reduced by taper and aspect ratio but more distributed

Form Drag



Form Drag is due to the shape of the wing

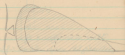


Form drag can be reduced by streamlining the shape. Skin friction the layer of air in contact with the wing is at rest and the next layer has a small velocity. Successive layers have increasing velocity until the friction of the air surface has no effect. The drag of the surface depends upon the thickness of this layer. It is more important for streamlined shapes and high speeds. It can be reduced by using smooth surfaces but it can never be completely eliminated.

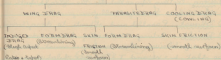
Parasite drag This is due to all parts of the aircraft that give no lift e.g. undercarriage, radiator, gunnery, engines, doors and landing. This drag can be reduced by streamlining the shape. The skin friction can be reduced by using smooth surfaces.



Cooling drag The airflow is used to cool the engine or radiator. This disturbs the airflow and so causes drag. It has been reduced by cooling. The heat of the engine may be used to attract the airflow giving negative drag i.e. the alternative waste energy from the engine in the form of heat is converted into useful work.



DRAG



Wing area and rise must be kept down to give less drag and still in cramped quarters for the crew. Speed by increasing the speed we increase drag.
D. $\propto V^2$

Therefore if we half the speed we quarter the drag so that reducing the drag saves the engine.
Straight and level flight

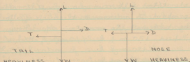


There are four forces acting on the aircraft in straight and level flight.

- ① Lift. Acting at right angles to the flight path in this case acting vertically.
 - ② Drag. Acting parallel to the flight path in this case horizontally.
 - ③ Thrust. Parallel and opposite to drag.
 - ④ Weight. Acting vertically downwards.
- For equilibrium these 4 forces must balance so that $L = W$ and $T = D$. The pilot unconsciously causes $L = W$, $T = D$ by manipulation of the throttle and controls. If $T = 10$ and $W = 10,000$ then L must equal 10,000 lbs D is 1,000 and

T must be 1,000 lbs to maintain straight and level flight.

C. of P and C. of G. These four forces must act in correct places or the aircraft will not be in equilibrium - may result in nose heaviness or tail heaviness.



These are called by the designer so the C.G.P. is behind the C. of G. in most land planes and the $L = W$ couple is balanced by the $T = D$ couple. When the thrust is cut out the aircraft is nose heavy and so takes up a gliding position.



Alterations of Speed. The lift must equal the weight, if the speed is reduced the total lift must be made up by increasing the lift from the angle of attack. This is obtained by increasing the angle. It can only be increased up to stalling angle when the total lift can no longer be maintained. If the speed is increased the angle of attack must be decreased to maintain equilibrium.

Example Speed	120 m.p.h.	Angle of attack	4°
	150 m.p.h.	"	3°
	220 m.p.h.	"	2°
	300 m.p.h.	"	1°

Stalling Speed 60 m.p.h. Stalling angle 16°

NOTE This only applies to one particular aircraft the speed applicable varies with all types.

When the stalling angle is reached the aircraft tends to get out of control - the pilot cannot measure the angle of attack, so he calls it stalling speed this stalling speed varies for different conditions of flight. The stall is due to loss of lift because of turbulent airflow and reduced speed. Landing is stalling the aircraft when very near the ground, so a low stalling

speed means a low landing speed.

Newton's Laws of Mechanics

19th Dec 1945

1. Moving air will tend to remain moving, still air to remain still, and to resist any change of direction or speed.
2. Force must be applied to the air to change its state of uniform rest or motion or direction of flow. The more sudden the change or the greater the air mass affected the greater must be the force applied.
3. Application of such force will cause an equal and opposite reaction on the surface producing it.

Boddy resistance

That portion of the resistance caused when air flows past a solid body disturbing the smooth streamline flow.

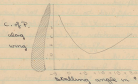
Advantages of cambered aerofoil

1. Less drag (smooth airflow)
2. Better lift (Venturi effect of bottom)
3. Depth gives strength

Centre of Pressure

That point on an aerofoil through which passes the resultant of all the lift forces acting on the aerofoil.

Centre of pressure moves at varying angles of attack due to the shifting pressure distribution. This causes instability, but modern design has largely overcome this.

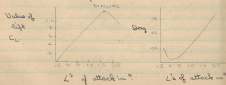


Lift and drag

Both depend on (i) airfoil shape and angle of attack (ii) air density (iii) airspeed (iv) wing area.

$$\text{Lift} = C_L \frac{1}{2} \rho V^2 S$$

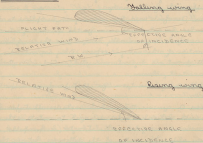
$$\text{Drag} = C_D \frac{1}{2} \rho V^2 S$$



Stalling speed

A certain minimum air speed needed to maintain lift below which speed further increase of angle will not maintain lift. However a wing stalls at an angle not a speed. Hence a high speed stall. A given lift can be obtained by either high speed & small angle or low speed & large angle.

X Autorotation X



Definitions

Adiabatic. The word which is applied in the science of thermodynamics to the changes which may take place in the pressure and density of a substance when no heat is communicated to it or withdrawn from it.

Acrobatics. Evolutions voluntarily performed with an aircraft other than those required for normal flying.

Aerofoil. A structure analogous to the wing or tail of bird designed to obtain reaction from the air at fixed or right angles to the direction of its motion.

Aeroplane. Any mechanically driven heavier-than-air aircraft with fixed wings.

Ailerons. Aerofoils used for causing an aeroplane to roll about its longitudinal axis for the purpose of balancing. These aerofoils are usually hinged near the wing tips in continuation of the wing surfaces. They are rotated in the opposite directions by interconnected control wires attached to the control column.

Aircraft. A generic term for all types of air-supported vehicles.

Air screws. A generic term which includes all types of screws with helical blades designed to rotate in the air. An air screw may be either tractor or pushpeller.

Airspeed ratio. The ratio of the difference between the

forward speed of an aeroplane and the forward speed which is necessary for the air screw to give a given thrust (i.e. no thrust). The air screw is considered to be rotating at the same number of revs. in each revolution.

Aspect ratio. The ratio of span to chord of an aerofoil.

Bank. The angle between the lateral axis of an aeroplane and the horizontal is called the angle of bank and is denoted by the symbol β .

Banker. The curvature of a surface of an aerofoil so bent, so tilted, so taken on inclined position.

Centre of gravity. That point in a body through which the resultant of the weights of its parts passes, in every position that the body can assume.

Chord. A straight line, touching the under surface of an aerofoil at or near the leading and trailing edges. The length of the chord is the projected length of a section of a plane i.e. normally the length of a line joining the leading and trailing edges. In an aerofoil having compound curvature the chord requires special definition.

Control column. The lever (or the pillar supporting a hand wheel) by which the elevator and aileron controls are operated.

Contributed forces. P. 1. 26. 74

Dihedral angle. In geometry - the angle between two planes. The planes of an aeroplane are at a dihedral angle when both fore and aftward are inclined upwards or downwards to the transverse axis. If the inclination is upward the dihedral is positive, if the inclination is downward it is negative.

Drag. The resistance along the line of flight, the head resistance i.e. the total resistance of the whole aircraft, including wings, fuselage, undercarriage etc.

Drift. The more sideways or sideways over the ground. It is measured as the angle between the course and the track made good.

Elevator. The horizontal control surface by which the upward or downward inclination of an aeroplane in flight is controlled. The elevator is usually hinged to the trailing edge of the tail plane.

Empennage. Tail unit

Endurance. The maximum time an aircraft can continue flying under given conditions without refuelling.

Fairing. An outer covering of the main frame structural control surfaces and fuselage. It is generally made of linen and kept in place by glue and is used to give the weather.

Fairing. Addition to any structure to reduce its head

resistance.

Fin. Vertical surface or surfaces forming part of the tail unit by which the turning of an aeroplane in flight is controlled.

Gliding out. The change of the path of flight from the gliding angle to a path parallel to the ground.

Gutter. An unstable oscillation due to the interaction of aerodynamic and elastic forces upon the inertia of any structure.

Gliding position. The attitude of an aircraft when the lateral and longitudinal axes are parallel to the ground.

Fuselage. The main structure or body of most types of aeroplanes, to which the main planes, tail unit and other surfaces are attached.

Ground speed. The speed of an aeroplane in flight over the ground.

Gap. The distance between one plane and the next immediately above or below it.

Incidence crapping. The angle between the chord of the main plane and the horizontal when the aircraft is in the crapping position.

Indicated horse power. The power actually developed in the cylinder.

Keel surface. The sum of all those areas on the aircraft which give rise to resistance forward when the aircraft slips leading edge. The forward edge of a streamlined body or aerofoil.

Lateral. Away from the wind.

Drift. The same as "lateral".

Lift. A force, perpendicular to the path of flight of an aircraft, which sustains the aircraft in the air.

Performance. The flying characteristics, such as the speed of level flight, rate of climb and ceiling, of an aircraft under specified conditions.

Distance flown. The distance through which an aircraft must advance along its own axis during one revolution in order to give no thrust.

Rolling. Angular motion about the lateral axis.

Plane of symmetry. A plane dividing an object into two parts each of which is a mirror image of the other.

Power. The time rate of expending energy in doing work.

Range. The maximum distance an aircraft can travel under given conditions without refuelling.

Rolling. Angular motion about the longitudinal axis. A complete revolution about that axis is commonly called a roll.

Side slipping and skidding. Movement sideways through

the air from the course ahead.

Thin friction. The resistance caused by the movement of a fluid (air) over the surface of a body.

Slipstream. The stream of air that is discharged aft by a revolving aerofoil.

Span. The overall distance from wing tip to wing tip.

Spinning. A diving descent combined with a continued rotation about the direction of motion of the centre of gravity of an aeroplane.

Stability. The quality by virtue of which any disturbance of steady motion tends to decrease.

Stagger. When main supporting surfaces are of equal chord, and when one such main supporting surface of a biplane is in advance of another these planes are said to be staggered.

Stalling speed. The airspeed corresponding to the maximum lift coefficient of an aerofoil. An aircraft is said to stall when its airspeed drops below that necessary to support it in the air.

Streamlined body. A body so shaped that it offers a minimum of resistance to motion thro' a fluid.

Thrust. The force acting along the axis about

which an airman is being rotated.

Time lag is the lapse of time between cause and effect.

Stall The moment of resistance of an airman to rotation.

Trailing edge The after edge of a streamline body or aerofoil.

Wash in An expression used to denote an increasing angle of incidence towards the wing tip.

Wash out An expression used to denote a decreasing angle of incidence towards the wing tip.

Wing The word used in aerodynamics to denote that portion of one of the main supporting surfaces of an aeroplane from the centre line of the a/c. to the wing tip.

Yaw To turn an aeroplane in the air about its normal axis. An aeroplane is said to yaw when the fore and aft axis turns to starboard or port out of the line of flight. The angle of yaw is the angle between the fore and aft axis of the aeroplane and the instantaneous line of flight.

Autorotation (cont)

Assume aircraft flying stalled hits bump causing one wing to rise, other to fall then falling wing meets relative upward airflow, angle of attack increases wing more stalled - less lift, wing drops even faster. Rising wing meets relative downward airflow - angle of attack decreases, wing now unstalled, now lift wing rises even faster. Result - a rolling moment in direction of initial movement which continues. Trying to raise the dropping wing by lowering its aileron as in normal flight has adverse effect.

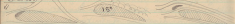
Dihedral (see definition). It gives lateral and directional effect.

Gap and stagger (see definition) normally gap equals chord. Stagger stops interference between the pressure areas, the amount being governed by pilot's view and accessibility to cockpit.

The ideal aerofoil. It would have a small moment of C. of P. @ high maximum value of lift (C_L) - low landing speed. @ low minimum value of drag (C_D) - high top speed. @ good L/D ratio - general efficiency and weight carrying. @ Depth of section - strength. Impossible to combine the above.

qualities in one wing. The three main types of aerofoil are high lift, high speed and general purpose. The main changes are in camber and depth. The large camber gives good lift but more drag therefore less speed. Small camber gives less drag therefore more speed but less lift.

Flaps.

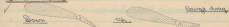


Small auxiliary aerofoils fitted by hinges parallel to leading edge. At normal angles they lie flush with the leading edge and have no effect. At large angles they are pushed out or about and obtain smooth airflow over wing via venturi effect by gap provided. Usually fitted along wingtip only for added lateral stability and control at low speed. Effects: 50-100% more lift, stalling angle delayed, stalling and landing speeds lowered, speed range increased, auto-rotation less likely. Aileron drag and control at slow speed.

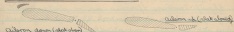
Bank is applied by increasing effective camber of one wing by lowering aileron. A secondary effect is that more of aircraft tends to swing towards turning wing tip. Lowered aileron being

more exposed than wing one to airflow and causing more drag. See fig. 35, sheet 1. For various types of aileron used to increase aileron drag. Considers a gliding turn to left at too slow a speed. Left rudder has little effect (no lift - slow outflow). Applying bank probably stalls rising wingtip. Aileron drag swings nose towards rising wingtip. Probable result is spin to right instead of turn to left.

Down aileron



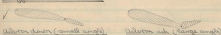
Interconnected flap and aileron



Automatic flap and interceptor



Differential aileron



Aerodynamic Balance

It is securing the airflow to assist pilot to most controls to save fatigue (see Lr 4/6).



Requires less than one fifth of total area in front of hinge.

Mass Balance & Static Balance



It is used to stop control flutter at high speed, due to pressure variation or bending moments. It also overcomes the effect of bumps (see Lr 4/6). It is sometimes concealed in front of the wing hinge.

Trimming tabs and Balance tabs (Lr 4/4)



They enable the pilot to trim away the pilot

in any aircraft in flight and today are combined with balance tabs.

Autocrew

Modern high speed aircraft with fixed pitch air screws have very poor static thrust i.e. poor take-off due to large blade angles necessary for the high speed in level flight. Fine pitch or small blade angles are necessary for good static thrust i.e. good take-off. The V.P. air screw is now used; it can be set by the pilot to fine pitch for take off and converted by him in the air to give the same relative blade angles at the higher speeds.

Forces acting on an aircraft in level flight

An aircraft is in equilibrium when flying:—

1. at constant height.

2. at constant speed.

3. in a fixed direction.

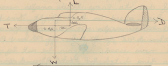
① occurs when lift equals weight.

② occurs when thrust equals drag.

③, which is prevention of rotation, depends on how the four forces are disposed.

Ideally all four would work through one point but C.P. and C.G. move in flight (change of angle

of attack or load). This forms a couple between L and W causing the nose to rise or fall. The usual arrangement is as indicated; a deliberate couple is arranged between L and W balanced by another couple $T D$.



Non lifting tail planes.

A tail plane is fitted to balance the forces under certain conditions. It must be symmetrical so as to give no lift when forces are balanced and when nose tends to rise or fall due to movement of c.g. or c.a. the airflow striking the tail plane will stop aircraft diving or trying to stall.

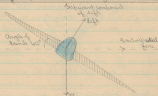
Lifting tail planes

Sometimes a permanent up or down load must be carried by the tail plane in flight e.g. certain flying boats (inverted cambered aerofoil with positive raggings incidence)

Adjustable tail plane

Enables pilot to adjust fore and aft balance of aircraft in flight.

"Forces acting upon an aircraft in a turn"



Aircraft flying S and L is in equilibrium. Therefore when turning it is no longer in equilibrium. When flying S and L it is obeying Newton's 1st. law and resisting any change of direction. Therefore an acceleration force known as centrifugal force must be applied towards center of turn to make aircraft turn and this is done by banking. Wing loading increases on turn and stalling speed rises. Therefore open throttle to overcome

loading and increase speed to make up for lift lost by banking.

Question

Aircraft is turning at 60° bank. What is the centrifugal force and the load factor if the aircraft weighs 10,000 lbs?



$$\text{load factor} = \frac{AB}{BC} = 2$$

$$\text{Centrifugal force} = AB = 10,000 \times \sqrt{3}$$

Motions of a turn

Bank and rudder in required direction, ease stick back to make aircraft turn. When desired bank is reached hold off bank (factor outer wing gets more lift and tries to go on banking).

There is a correct degree of bank for every

radius of turn, the quicker the turn, the steeper the bank, because the greater is the C.F.F. medium. Overbanking or under banking cause slipping in and skidding out

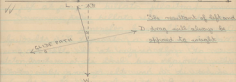
Sliding turn

Increase speed to overcome loading, then proceed as above, little or no need to hold off bank. Outer wing faster but inner wing steeper spiral downwards.

Slipping turn

Proceed as above but must hold off considerable bank (outer wing both faster and longer spiral upwards).

Forces acting on an aircraft when gliding



Modern aircraft have flatter glide paths than older types because better design gives better $\frac{L}{D}$ ratio

Flaps

When lowered, camber is increased therefore lift and drag also. Pull them down for landing, not more than half down for take off (max lift for least drag)

Steeper glide path, lower landing speed, act as air brakes on landing run, enable fast aircraft with flat glides to land on aerodromes with obstructions on boundary, did otherwise they could not use

Spinning

A form of autorotation due to applying rudder at the stall. Movement occurs in all three planes around the spinning axis with one wing partially stalled trying to fly around the other stalled wing

Axis and planes of reference

Normal axis is a straight line through C.G. at right angles to the other two axes. When straight and level it is vertical, rotation around it is in the yawing plane controlled by rudder. Longitudinal axis (rolling plane) and lateral axis (looming or pitching plane)



Loading during a loop



Trimming tabs

1. Fixed tab consists of a strip of metal fixed to the trailing edge of the main control surface which can only be adjusted by hand on the ground. It is used to correct slight rigging faults
2. Controllable type. Controllable in flight by the pilot from the cockpit to correct

in manoeuvre to assist in take-off & give trim for hands and feet off control when fitted to all control surfaces. & it obviates the necessity of fitting an adjustable tail plane.

3. Automatic or balance tab is a form of aerodynamic balance and is fitted in such a way that when the main control surface moves the tab moves in the opposite direction.

4. Combination of 2 and 3

Rudder bias sometimes used instead of or in conjunction with rudder trimming tabs. It is a spring loading of the rudder bar, controlled by the pilot to:- a. overcome effect of slipstream b. reduce yaw with one engine c. give trim for straight flight, feet off.

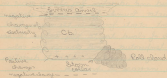
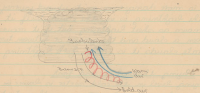
Hail is more dangerous on its upward journey. If the lightning is in the cloud itself it is known as sheet lightning but if it occurs between a cloud and the ground or between two separate clouds it is called fork lightning.

Flying through a cold front

Freezing level in the cold sector is lower than in the warm sector. Muggy weather, strati formed cloud and advection fog always occur in the warm sector. Cumuli formed clouds always in the cold front, strati formed in the warm front. Heavy bank of cloud, try to fly between the clouds. Decrease of temperature, wind changes from calm breezes to squall wind in opposite direction. Wind veers, increase in pressure.

Descent sandstorm

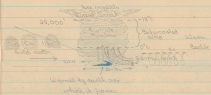




Hazards encountered in a cold front

1. Violent turbulence
2. Ice accretion
3. Hail
4. Lightning and thunder
5. Heavy rain
6. Bad visibility

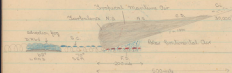
Cold Front



Line squall

A line squall is a series of thunderstorms along a well developed cold front, when there is a lull in the upward currents cold air descends and is pushed forward giving a line squall.

Cold front



The base of the Oc. is often higher than that of the N.S. This is because the surface over which the Oc. forms is warmer than the cold air over which the N.S. forms. If on entering a warm front and icing is encountered, turn back, climb and use de-icing equipment.



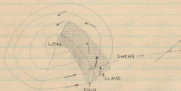
Cold front

Warm front

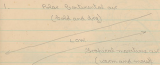
Oc. is a lift of warm air on surface

The speed of the depression is the same as the direction and speed of the wind in the warm sector. When the cold front catches up to the warm front the result is an occlusion.

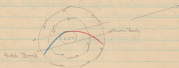
Warm front



Air masses, depressions and fronts.



2. Cyclones



The cold air being denser than the warm, the friction caused is greater and thus the cold air movement is slower than the warm. Warm air over-rides the cold air on the warm front and cold air forces its way underneath the warm air thus causing the warm air to rise giving the cold front. The stronger the depression the faster it moves.

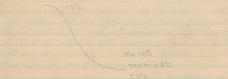
Glazed frost icing.

Forms in falling rain which comes into air below freezing point. As it hits the aircraft it spreads out and is frozen. The longer the drops the more the accumulation.

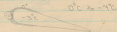


Sublimation

When an aircraft, descending rapidly from a height where temp. is say -10°C , comes into air of temp. about 5°F ice will form on the aircraft. This is due to the air in contact with the cold surfaces of the aircraft being cooled below its dew point and F.P. thus causing the water droplets produced to freeze.

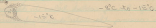


Transparent or translucent



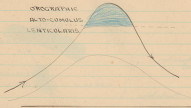
Water droplet hits wing. Part of it freezes. In doing so gives out latent heat which causes the water to spread out over the wing. The water, being warmer than the wing, heat will flow from it to the wing thus reducing the water below its freezing point. The ice builds up rapidly and freezes up moving faster. Occurs in low layers of cloud.

Rime icing



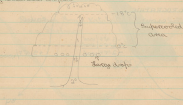
This forms on the leading edges when in the high layers of cloud. It interferes with the airflow but breaks off in lumps. The less the temperature, the less the drops spread. The smaller the drop the less it spreads. Forms on aircrews and may thus cause vibration besides damage to the airframe when it breaks off.

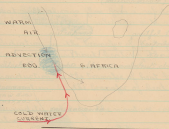
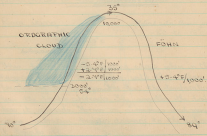
OROGRAPHIC
PL. TO-CUMULUS
LENTICULARIS

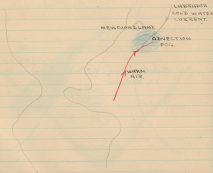


Ice accretion

Very small water droplets







S. F. T. S.

Heat will always flow from a substance with a higher temp. to a substance with a lower temp.

Radiation is the transfer of heat from one body to another without heating the medium through which it passes.

Conduction is the transfer of heat by contact.

Convection is the transfer of heat bodily.

Humidity refers to the moisture content of the atmosphere.

Specific humidity is the actual amount of water vapour measured in gms per kgm.

Relative humidity is the ratio of the actual amount to the total amount required to saturate the air at that temperature expressed as a %.

Dew point is the temp. to which air must be cooled in order to become saturated.

Cold Front

The following changes are experienced:-

- a. Large C. & C. Heavy rain
- b. Decrease in temperature
- c. Veering wind
- d. Followed by fair weather and good visibility.

Polar Air Mass

- a. Is unstable if it moves into warmer regions. Produced by the heating from below.
- b. It will thus give C. thunderstorms and a fair amount of rain.
- c. Good visibility owing to particles being carried up by the vertical currents.

Tropical Air Mass

- a. Is stable if it moves into colder regions.
- b. Will give low cloud if it moves into colder regions, and drizzle, if its lower layers are cooled below dew point.
- c. Poor visibility due to particles being kept in lower layers (No convection currents).

Warm Front

With the passage of a warm front the following weather changes are experienced-

- a. Continuous precipitation.
- b. Increase in temperature.
- c. Veering wind.
- d. Fog and low cloud or fair weather later.
- e. Usually followed by poor visibility.

	TROPICAL		POLAR	
	CONTINENTAL	MARITIME	CONTINENTAL	MARITIME
Temp.	High	High	Low	Low
Humidity	Low	High	Low	High
Pressure	Low	Low	High	High
Weather	Clear	Overcast, fog, muggy	Clear	Shallow cumulus, etc.
Cloud	Clear to	Sc. and bc.	Clear to	bc. to
Visibility	Heavy	Poor	Good	Good

STABLE

UNSTABLE

Weather Observations

1. Surface observations made at each hour (and also when sudden changes occur) at met. offices and other places such as lighthouses, observatory posts, etc. in the U.S.
2. Observations called plots, made by aircraft crews during flight and recorded in on landing.
3. Observations made by a number of special met. flights (temperatures, wet and dry bulb at various pressure levels).
4. Observations of upper winds.
5. Observations from Allied and some neutral countries

Stable Air

It is air that having been forced to rise, will only rise to a limited height and will then stop or tend to return to its original position. Results in fair weather with little or no cloud.

Unstable Air

It is air that having once started to rise will continue rising indefinitely. Produces large masses of cloud with thunderstorms and rain.

Stable

Unstable

			Rising Air		
	65		55	54	
Rising Air	66	upper			upper
	67	air	60	58	air
		temp	65	62	temp
↑ 65	68				
40	69		40	66	
45	40		45	40	

Duties of the met. officer

1. To make met. observations
2. To supply routine weather forecasts and warnings
3. To provide weather forecasts for all intended flights
4. To describe weather, past, present and future, accurately and promptly.

It is ^{not} the duty of the met. officer to pronounce upon the suitability of the weather for flying. This is done by the C.T. officer authorising flights after consultation with met. officers.



REAR OF FORM 2330

(Observations by crew of aircraft)

TIME	LOCATION	DATE	TIME	LOCATION	DATE	TIME	LOCATION	DATE	TIME	LOCATION	DATE
0730	04°40'N 100°00'W	-11	0800	04°40'N 100°00'W	-11	0900	04°40'N 100°00'W	-11	1000	04°40'N 100°00'W	-11

Form 2330

This is a route forecast and not a weather report. It is issued as a matter of routine on all operational flights and route forecasts should be obtained for any flight over 100 miles (A.M.C.). Ample warning should always be given to the met. officer if route forecasts are required stating route, time of take-off, and duration of flight. Heights as all A.M.C.L. Observations are made by crew during flight therefore these forms should be returned immediately on landing as they are used for plotting on a synoptic charts, called Plots.

Form 2331 Index to code figures.

Dry Adiabatic Lapse Rate equals 5.4°F per 1,000 ft

Saturated Adiabatic Lapse Rate equals 3.7°F per 1,000 ft

Lapse Rate

As pressure decreases with height in the atmosphere at a constant rate. If air is forced to rise, it enters an area of reduced pressure and expands which uses up energy. This energy is in the form of heat, so that rising air will lose this heat (adiabatic condition) and a drop in temperature will therefore take place.

56	56	
58	58	58
	D.A.L.R.	
61	61	61
		Temperature in surrounding air
64	64	64
	D.A.L.R.	
69	71	69
70	76	70

Rising Air

Types of weather information available at met. office

1. Route Forecasts

Available at any time on demand (notice should be given). Based on Form 2330

2. Local forecasts

Based on a matter of routine, say three times a day, and only apply to a radius of 20 miles around the station. Used as a guide for local flying.

3. Typhal warnings

4. Squall and thunderstorm warnings. Movement and development of storms are watched and a warning given of the arrival at a certain place.

5. Fog and fog warnings

6. Barometric pressure

7. Warnings of sudden deteriorations or changes in the weather to air in flight

8. Time of sunrise and sunset.

Form 2330

STAGE	Outward and thrustout	Surge	Barometer
SURFACE WIND			Windspeed
2,000'	300'/11 mph		
4,000'	270'/10 mph		
6,000'	230'/8 mph		
	mainly light squalls	fair	fair
WEATHER	for isolated showers		
VISIBILITY	good except in showers	heavy	good
CLOUDS	8-75% at 4,000' 6-85% in showers	Sea at 4,000'	Breakers 100'
ICING	High in Cn above		
ICEBERG	10,000'		
FREZZING	8,000'		
LEVEL			
PRESSURE	1		LOW mb. 98%
STATIONARY			change
PRESSURE AT			1014-1016 mb.
H.S.L.			change
REMARKS	Condensation bands likely above 10,000' Heavy swell on sea		

Types of mt. stations

Type 1 attached to group of command H.Q. and does all administration & constructs synoptic charts and issues general forecasts at any time. Acts as receiving centre for collection of reports (e.g. Salisbury)

Type 2

Situated where information required cannot be met by Type 1 & is large bomber station. Constructs synoptic charts and issues independent forecasts. Prep. brief. issue form 2330, and observe (e.g. Bulawayo)

Type 3

Constructs synoptic charts but receives forecasts and brief and issue form 2330 to aircrew. Situated at bomber stations (e.g. Kumbani & Bloemfontein)

Type 4

Simply act as a link between some mt. office and aircrew. Do not construct charts but observe (e.g. Dordrecht). Found mostly at fighter stations

cloud illustrations Form 2330

Aviation weather report 4th Oct 43

An aviation weather report consists of the first three groups of the international code. All observations are visual, and thus it is simply a statement of the actual weather at a certain place at a certain time.

III C L C N. W W V h N h. D D P W N.

These observations are made at regular intervals at all reporting stations. They are transmitted by the quickest method to a collecting center where they are used for making route forecasts and for plotting on a synoptic chart.

Pressure and temp are reduced to a common level before plotting.



Effects of air change of density on an alt

To summarize in general terms, at low levels (where air is dense), lift is high, air seems slip to less, engines are more powerful and airspeed indicators read correctly. Conversely, at high levels (where air is thin) lift is lower (hence an aircraft finally reaches its ceiling), engines are less powerful (hence the introduction of superchargers to compensate) and airspeed indicators read low. Decreased air density at mountain conditions greatly increases take-off and landing runs.

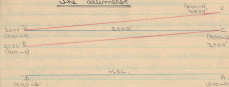
Weather Reports

Bulawoye - large cumulus or cumulonimbus cloud and thick altostratus or nimbostratus heavy thunderstorms with rain or snow, vis. between 500 and 1000 metres. Base of low cloud 200-300 metres high about 1/2 of sky covered by it. SE wind fresh breeze. Then has been showers and sky is overcast. Later, sky is cloudless, visibility up to 100 miles. No cloud below 3,000'. Calm and fine Salisbury. Fair weather cumulus with no rain.

Information put on weather maps

- 1, Atmospheric pressure reduced to M.S.L.
- 2, Barometer tendency (change in pressure in the last three hours)
- 3, Temperature
- 4, Humidity (usually dew point)
- 5, Visibility
- 6, Clouds (type, amount of low cloud, height of base, total amount of cloud)
- 7, Weather (present and past)
- 8, Surface wind in direction and force.

The altimeter



N.B. Pressures in brackets have been calculated by means of the 30' mb rule and are therefore only approximate.

A. Hemispheres

Port wind (starboard drift) Altitude correct
Starboard. Port drift. Underread

B. Hemispheres. Vice Versa

30 mph wind on beam altimeter will be in error by 1 ft. for every mile flown

If actual mean temp. of air column is above the low value for the mean temp. then altimeter will under-read.

Visibility

Visibility is the furthest distance at which objects are clearly recognizable in a horizontal direction by a ground observer with the naked eye

If visibility is 400 - 2500 yds it is called mist or haze
.. less than 100 yds .. fog
.. less than 250 yds it may be dust storm

It is measured at night by using a light of specified candle power and viewing it through smoke glasses of different shades. Each of these shades represents a certain visibility at night

Relative humidity is the ratio of the amount of water vapour present in a given volume to the amount which that volume could contain if it were saturated, expressed as a %

Measuring the height of clouds

By day

1. By using a pilot balloon, the time is noted of the time the balloon takes to disappear.

2. By means of an altimeter

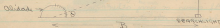
3. By judgement from practical experience.

4. By comparison with hills or mountains, the heights of which are known beforehand.

At night a Chinese lantern may be fixed to a pilot balloon but this method is seldom used. The most common method used for finding height of cloud at night is by having a searchlight placed a definite distance from the met. office. It is pointing vertical and the angle from the met. office and where the searchlight hits the cloud by means of an alidade.

By night

Height of cloud is 3.2 km



Stratus at Sea

It is formed by cold air passing over a warmer sea surface. Well out to sea there is no marked variation. Near to land it tends to form in the early morning but as soon as the land gets warmer it is dispersed.

Condensation Trails

The condensation of engine exhaust water vapour sometimes causes trails behind aircraft flying in clear air. They may be faint and invisible from any great distance; on the other hand they may be long white streaks betraying aircraft position and lasting and spreading out for twenty minutes or more. Usually they may look more like cirro-cumulus, alto-cumulus or cirrus cloud.

Dense trails rarely form under about 22,000' in summer and 22,000' in winter. They always cease on, or shortly after, entering the stratosphere. Dense trails may therefore be avoided by climbing after flying. Otherwise the pilot should seek dense air, usually found just above a cloud layer. If cirrus or any other high cloud is about beware of the trails.

Operational use of clouds

1. Direct use

- 1.1. for tactical purposes.
- 1.2. for cover in attack for cover for evasion

Important points

1. Degree of continuity
2. Height of base
2. Vertical thickness
4. Visibility
3. Turbulence
4. Being.

2. Indirect use

Clouds often give good indication of weather to be expected in the near future or in certain direction.

A frequent sequence of clouds is as follows.

Cirrus - Cirrostratus - Altostratus - Nimbostratus

It is caused by a warm air mass forcing over a cold air mass.



If in A.S. the clouds just clear A.P. and below it is just clear. Below A.P. clouds may vary.

Name	<i>Leptocarpus</i> (No.)	<i>Leptocarpus</i> (No.)
Description	Brown, dense, white center of leaf against blue sky. No color.	Thin, white, not very dense. No color, or very faint. No color, or very faint.
Use of wood	20-30,000'	20-30,000'
Vertical thickness	Thin	Thin
Disability	Very good	Very good
Transparency	Smooth	Smooth
Permeability	Nil	Nil
See section	Nil	Nil
Suitability for single or flying	No danger	No danger
Suitability for formation flying	No danger	No danger
Direct	Nil	Nil
Operational Value	Too high and too thin	Too high and too thin
Indirect ops. Value	After indirect climbing value. Indicate	Indicate, depending on value. Indicate
Additional notes		

<i>Leptocarpus</i>	<i>Leptocarpus</i>	<i>Leptocarpus</i>
Range of very white flowers of colorless leaves in form of waves resembling ripples	Range of fairly small colorless flowers in waves, large number of waves, large number of waves	Range of small, usually colorless, many flowers of waves, large number of waves, large number of waves
20-30,000'	4-20,000'	4-20,000'
Thin	Thin	Thin to 10-15,000'
Very good	Good	Thin 100-15,000' up thick 20-100,000'
Smooth	Smooth	Partly smooth
Nil	Nil	Occasionally thin
Nil	Unlikely and harmless	May occur but not likely to be serious
No danger	No danger	Suitable
No danger	No danger	Usually suitable
Nil	LITTLE	Very good, very good, very good
Too high and too thin	Looks continuous, but not too good, strong discoloration, very good	Good for vision (but cloudy of all)
Indicates some good formation trials	Generally little	Indicates off, much of wave form.
Very, very often colored with red	Red and red sometimes faded, marked, very	

<i>Microtus</i> No.	<i>Microtus</i> No.	<i>Microtus</i> No.
Uniform dark gray large expanding white. May with S.E. beneath.	Large of partly large expanding material in lines or waves often giving together	Uniform hoodless large narrowly gray. May have enveloping line like
1. - 7,000' or lower	3. - 4,000'	Less than 1,000'
10. - 15,000' or more	500 - 3,000'	300 - 2,000'
Red often beneath wings	10. - 20 generally, 12 - 20	usually less than 10 yds or less - variable
Red often above faint, dark, more often faintly continuous may be partly heavy & the hoodless of light good hoodless	Brilliant smooth hoodless fine dotted may occur but not usually heavy (0% - 7%)	Sometimes spotty sometimes dotted only in winter not heavy (0% - 1%)
Fractured only if adequate change is maintained	Fractured	Dangerous.
Totally impracticable	Fractured for small up but only if no C.C. or C.C. is present.	Impracticable
Good can be dangerous	Good, good even and good for occasion.	Can fly above it only served for surface. may use more area
Part of warm front cloud sequence	Little	Little
Flying beneath the is changes, rising to but variability and the enveloping hills	about the large flying is very smooth. May form be warmer & large, and may be different in direction & strength	Indistinguishable may form fog from above

<i>Microtus</i> No.	<i>Microtus</i> No.	Description
Detached back with flat base & marked top of considerable vertical development.	Heavy increasing mass with faintly multi-faceted top of spreading out like in mass.	Description
2. - 3,000' ft.	3,000 ft. or lower	Alt. of base
4. - 5,000' or more	12,000 - 20,000'	Vertical thickness
Less often beneath wings	Red often less 10 yds	Visibility
Red often more	Direct	Surroundings
No precipitation	Heavy Rain, Red, snow, dust.	Precipitation
May be heavy and rapid (0% - 15%)	May be very heavy and rapid (0% - 15%)	Accretion
Fractured	Dangerous.	Suitability for single th. flying
Totally impracticable	Totally impracticable	Suitability for formation flying.
Backs continuously red	Dangerous but good good for occasion.	Direct operational Value
Must contain indirect value associated with direct observed variation.		Indirect operational Value.
	Lighting is likely in S.E. winds must be ward in and with the must be formed (all hairs in water, but heavy cloudy, dark)	Additional notes.

on below us. May form over hills. Great vertical extent. Visibility is bad. Bumpiness occurs and may be severe. Rain occurs and may be heavy. Snow occurs. It is quite likely there may be ice accretion, may be thick. Flying is bad beneath us. Danger of hills. Very dangerous cloud for flying.

Stratus

Continuous layer, covers whole sky, not very thick, very low base, and ragged. Varies in height from 0 - 1,000 ft. Almost same as fog. Flying in it is very risky indeed. Does not produce rain. Icing, not common but when it does occur is very light. Not bumpy.

Altostratus

Between 2000 - 4000 ft. A layer of fairly large globules or ~~fine~~ rolls sometimes forming lines or waves. Layer may be from 500 - 2000 ft thick. Visibility variable, usually 15 - 20 yds. Fair to moderate as a rule. Not rain producing. Fairly smooth flying condition, above clouds it is very smooth. Icing is common but not likely to become serious. Only dangerous when temp 0° - -4° C. Wind direction above cloud may be different than below.

Cumulus, cumulo always signify heap cloud. Nimbus, nimbo rain producing. Strato cumulo or fractostratus... broken.

Flying conditions in clouds

Danger zone for icing in layer cloud is from 0° C to -10° C.

Stratus, cirrocumulus, and cirrostratus are ideal for flying but afford no cover from hostile aircraft.

Altostratus

Being unlikely. If it does occur it will not be very dangerous as droplets are very small and scattered. Good visibility. No bumpiness. No precipitation.

Cirrostratus

Generally covers whole sky. ~~Is~~ greyish. (It does not blot outlines of sun or moon). Variable in thickness. May blot or cut out sun or moon. Visibility is fair, good in thin cloud. Fairly smooth flying. May produce light rain but not usual. Icing may occur but is not serious.

Nimbostratus

Uniform dark layer. Low in sky. Rain or snow producing. Ragged misty low cloud may join

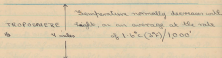
Altimeter Pressure Error

If we know accurately the relationship between pressure and height we can take an aneroid barometer and graduate the dial to read height and pressure. This instrument will then always record drop in pressure as increase of height and rise in pressure as decrease in height. Height is not the only factor which affects atmospheric pressure. In any particular place pressure may vary from hour to hour. Also pressures at places at the same level may vary across the country (horizontal variation). These horizontal variations are very small compared with those in rising the same distance vertically but nevertheless they are often big enough to affect the altimeter.

Imagine an aircraft flying level at 3,000'. If the atmospheric pressure is dropping the altimeter will register a change since it is really an aneroid barometer. Drop in pressure is usually associated with increase of height and so the altimeter shows a reading of more than 3,000 ft. although the aircraft is still at the same level i.e. it overreads.

their origin to unequal heating at the equator and the poles

STRATOSPHERE Temperature remains constant with height



Rate of decrease of temp. with height is called LAPSE RATE

Layers in troposphere in which temperature increases with height is called INVERSION



Formation of cloud and fog

Humidity At a given temperature a given volume of air can only hold a limited weight of invisible water vapour. When it holds this limited amount it is said to be saturated and at that temp. it can hold no more. The higher the temp. the greater the weight of water vapour necessary to saturate a given volume of air. We define the amount of water vapour present in air as the relative humidity which we may state as weight of water vapour present at given

weight of water vapour required for saturation

This is usually expressed as a %, e.g. if a cubic foot of air contains 1 gm. of water vapour and is saturated when 0.85 gm. of water vapour are added then R.H. = $\frac{1}{0.85} \times 100 = 80\%$

1-23

If an unsaturated mass of air be cooled it must eventually reach a temperature at which it is saturated and invisible drops of water condense out. This temperature is known as the dew point. Both relative humidity and "dew point" are measured by an instrument called the wet and dry bulb thermometer.

Cloud. For cloud to form air must be lifted sufficiently to cool below its "dew point". Cloud consists of visible drops of water plus industrial particles. When air is lifted and unsaturated it cools rapidly (B.H. & F. factor) due to the decreasing pressure allowing it to expand.

Clouds may be formed in three ways

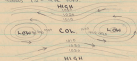
A. Mechanical lifting e.g. due to a range of mountains in a bath of moving air.

B. Thermal lifting due to unequal heating of the earth's surface causing convection currents.

C. Frontal lifting due to the meeting of a warm air mass and a cold air mass. In the first place these air masses are

system. weather fine as in anticyclone. Difference
fine weather rather brief (thanks to the low) due to
proximity of low pressure.

Feb. no visitors, no no wind or light and variable
winds. In winter fog for several days but no
ice with the low or disappears no weather
only days 12-24 hrs.



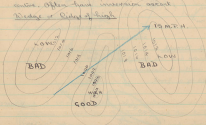
Good weather system

High pressure or anti-cyclone



Wind at 1000 ft given by the isobars clockwise.
at surface wind away from the center still toward
low pressure. Have descending air currents in the
center. After hour inversion absent

Wedge or ridge of high



no pointed isobars no front associated with this

and have now reached the stratosphere where the temperature remains constant.

Inversion

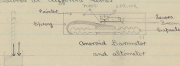


When temperature increases with height, often occurs on still clear nights when the land cools and so cools the lower layers of air. Cold air does not rise. The earth is considered as being built up of five zones - Tropics, Tropics, Temperate, Temperate, Polar.



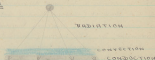
Density A decrease in pressure means a decrease in density. A decrease in temperature means an increase in density but only when the pressure is kept constant. Actually the decrease in density due to change of pressure overpowers the increase due to temperature change. Therefore an increase in altitude means decrease in pressure, temperature and density.

equal one millibar and 20,000 ft. Pressure varies horizontally, places at sea level having differing pressures at different times.



Aneroid Barometer

Radiation



- ① Heat transfer by radiation to the ground
- ② Ground transfer heat to the air next to it by conduction.
- ③ Warm air rising causes convection currents. Temperature decreases with height, the normal rate equals 3°F . for 1000 feet. At about seven miles the latter rate ceases. This is the end of the troposphere.

Meteorology

It is studied because ① it increases air crew efficiency and safety ② it enables air crew to understand the medium in which they work ③ it enables air crew to discuss ~~weather~~ information with the inst. officer ④ it enables air crew to bring back useful weather information.

Atmosphere

This consists of ① gases, oxygen, nitrogen and water vapour. ② liquid water ③ solids - water, dust, smoke

Pressure

The air exerts a pressure depending upon the column of air above it (compressive head of weight). This pressure is measured by a mercury barometer or an aneroid barometer. In the mercury barometer the air pressure is balanced against a column of mercury (about thirty inches at sea level). In the aneroid barometer the pressure is balanced against the elasticity of the lid of a hermetically sealed, partly exhausted box or capsule. Air-craft altimeters are aneroid barometers calibrated in feet instead of millibars. Pressure varies with height thirty feet being equivalent to one millibar. This is an average reading for medium heights, sixty feet

